



Airworthiness Directive

AD No.: 2019-0309

Issued: 19 December 2019

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS

Type/Model designation(s):

A318, A319, A320 and A321 aeroplanes

Effective Date: 02 January 2020

TCDS Number(s): EASA.A.064

Foreign AD: Not applicable

Supersedure: None

ATA – Aircraft Flight Manual – Amendment

Manufacturer(s):

Airbus, formerly Airbus Industrie

Applicability:

Airbus A318-111, A318-112, A318-121, A318-122, A319-111, A319-112, A319-113, A319-114, A319-115, A319-131, A319-132, A319-133, A320-211, A320-212, A320-214, A320-215, A320-216, A320-231, A320-232, A320-233, A321-111, A321-112, A321-131, A321-211, A321-212, A321-213, A321-231 and A321-232 aeroplanes, all manufacturer serial numbers.

Definitions:

For the purpose of this AD, the following definitions apply:

The AFM TR: Airbus Aircraft Flight Manual (AFM) A320 Temporary Revision (TR) 502 “Abnormal V alpha Prot”, issue 1.

Reason:

EASA AD 2015-0135 (currently at Revision 3) does not explicitly require removal of the AFM TR, as defined in this AD, which is intended to be removed following modification of an aeroplane to a configuration as specified in Table 3, Table 4, or paragraph (19), as applicable, of that AD. Whereas the benefit of applying that emergency procedure is confirmed on pre-mod aeroplanes, it has been determined that for post-mod aeroplanes, retaining the AFM TR in the AFM is not allowed. Due to



the absence of a clear requirement to remove the AFM TR, some aeroplanes may still be operated with that emergency procedure in the AFM.

This condition, if not corrected, could, under certain conditions, lead to wrong application of the procedure by the flight crew, possibly resulting in increased flight crew workload and consequent reduced control of the aeroplane.

For the reason described above, this AD requires, for certain aeroplane configurations, amendment of the AFM by removing the Emergency Procedure as specified in the AFM TR, as defined in this AD.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

Post-modification AFM Amendment:

- (1) Within 30 days, or concurrently with installation of an ELAC configuration as identified in Table 3 of EASA AD 2015-0135R3, or later configuration, and installation of an AOA sensors configuration as identified in Table 4 of EASA AD 2015-0135R3, or as identified in paragraph (19) of EASA AD 2015-0135R3, as applicable, whichever occurs later after the effective date of this AD, remove the AFM TR, or the copy of Appendix 1 of AD 2015-0135 (any revision), as applicable, previously inserted in the AFM as required by paragraph (8) of EASA AD 2015-0135, from the AFM.
- (2) Before next flight after amendment of the AFM on an aeroplane as required by paragraph (1) of this AD, inform all flight crews and, thereafter, operate that aeroplane accordingly.

Ref. Publications:

Airbus AFM A320 TR 502 issue 1, EASA approved 05 December 2014.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication.
3. Enquiries regarding this AD should be referred to the EASA Programming and Continued Airworthiness Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#).
5. For any question concerning the technical content of the requirements in this AD, please contact: AIRBUS – Airworthiness Office – EIAS; Fax +33 5 61 93 44 51; E-mail: account.airworth-eas@airbus.com.

