



Airworthiness Directive

AD No.: 2020-0096

Issued: 29 April 2020

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I, Part M.A.301, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I, Part M.A.303] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

PILATUS AIRCRAFT Ltd

Type/Model designation(s):

PC-24 aeroplanes

Effective Date: 13 May 2020

TCDS Number(s): EASA.A.594

Foreign AD: Not applicable

Supersedure: None

ATA – Airplane Flight Manual – Section Normal Procedures / Electronic Circuit Breaker – Amendment

Manufacturer(s):

Pilatus Aircraft Ltd

Applicability:

PC-24 aeroplanes, all manufacturers serial numbers (MSN).

Definitions:

For the purpose of this AD, the following definitions apply:

The AFM TR: Pilatus Aircraft PC-24 Airplane Flight Manual (AFM) Temporary Revision (TR) 02371-016 (“Change of ECB Check In Before Engine Start Procedure”).

Reason:

An occurrence was reported where, before take-off after maintenance of a PC-24 aeroplane, some electronic circuit breakers (ECB) were found in a “LOCKED” state.

This condition, if not corrected, could lead to a loss of power supply to equipment, without indication to the flight crew before take-off.



To address this potential unsafe condition, Pilatus issued the AFM TR, as defined in this AD, to provide operators with the necessary preflight check instructions.

For the reason described above, this AD requires amendment of the AFM.

Required Action(s) and Compliance Time(s):

Required as indicated, unless accomplished previously:

AFM Amendment:

- (1) Within 30 days after the effective date of this AD, amend the AFM by inserting a copy of the AFM TR, inform all flight crews and, thereafter, operate the aeroplane accordingly.
- (2) Amending the AFM to incorporate a later revision, which includes the AFM change as required in this AD, is acceptable to comply with the requirements of paragraph (1) of this AD.

Ref. Publications:

Pilatus PC-24 AFM TR 02371-016 original issue dated 01 November 2019.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. This AD was posted on 27 March 2020 as PAD 20-059 for consultation until 24 April 2020. No comments were received during the consultation period.
3. Enquiries regarding this AD should be referred to the EASA Programming and Continued Airworthiness Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.
4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#).
5. For any question concerning the technical content of the requirements in this AD, please contact: Pilatus Aircraft Ltd, Customer Support General Aviation, CH-6371 Stans, Switzerland Telephone: +41 848 24 7 365, E-mail: techsupport.ch@pilatus-aircraft.com, Website: www.pilatus-aircraft.com.

